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Aerodynamic Parameters Formulas

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List of 11 Aerodynamic Parameters Formulas

Aerodynamic Parameters

1) Sideslip Angle for Aircraft

$$\text{fx } \beta = \alpha_v - \sigma$$

[Open Calculator !\[\]\(a870788d6ed9b8fd294b7654a8c8526b_img.jpg\)](#)

$$\text{ex } 0.05\text{rad} = 0.117\text{rad} - 0.067\text{rad}$$

2) Sideslip Angle for given Moment Produced by Vertical Tail

$$\text{fx } \beta = \left(\frac{N_v}{l_v \cdot C_v \cdot Q_v \cdot S_v} \right) - \sigma$$

[Open Calculator !\[\]\(c50c8b7b2cc2cf9ff925edec0ee94c0d_img.jpg\)](#)

$$\text{ex } 0.049883\text{rad} = \left(\frac{5.4\text{N}\cdot\text{m}}{1.2\text{m} \cdot 0.7\text{rad}^{-1} \cdot 11\text{Pa} \cdot 5\text{m}^2} \right) - 0.067\text{rad}$$

3) SideSlip Angle for given Yawing Moment Coefficient

$$\text{fx } \beta = \left(\frac{C_n}{l_v \cdot S_v \cdot Q_v \cdot \frac{C_v}{S \cdot b \cdot Q_w}} \right) - \sigma$$

[Open Calculator !\[\]\(f60b7a900783ac3fd531bfd9c111be6d_img.jpg\)](#)

$$\text{ex } 0.04984\text{rad} = \left(\frac{1.4}{1.2\text{m} \cdot 5\text{m}^2 \cdot 11\text{Pa} \cdot \frac{0.7\text{rad}^{-1}}{5.08\text{m}^2 \cdot 1.15\text{m} \cdot 0.66\text{Pa}}} \right) - 0.067\text{rad}$$



4) Sideslip Angle given Yawing Moment Coefficient and Tail Efficiency

$$\text{fx } \beta = \left(\frac{C_n}{V_v \cdot \eta_v \cdot C_v} \right) - \sigma$$

[Open Calculator !\[\]\(cbe80b694ebd74fcfe136a095b608235_img.jpg\)](#)

$$\text{ex } 0.050694\text{rad} = \left(\frac{1.4}{1.02 \cdot 16.66 \cdot 0.7\text{rad}^{-1}} \right) - 0.067\text{rad}$$

5) Sidewash angle

$$\text{fx } \sigma = \alpha_v - \beta$$

[Open Calculator !\[\]\(3e2231b1ad3ca8da8658228c00dd08e0_img.jpg\)](#)

$$\text{ex } 0.067\text{rad} = 0.117\text{rad} - 0.05\text{rad}$$

6) Sidewash Angle for given Moment Produced by Vertical Tail

$$\text{fx } \sigma = \left(\frac{N_v}{l_v \cdot C_v \cdot Q_v \cdot S_v} \right) - \beta$$

[Open Calculator !\[\]\(0d5ec72f61334709c3fc9450209b754f_img.jpg\)](#)

$$\text{ex } 0.066883\text{rad} = \left(\frac{5.4\text{N}\cdot\text{m}}{1.2\text{m} \cdot 0.7\text{rad}^{-1} \cdot 11\text{Pa} \cdot 5\text{m}^2} \right) - 0.05\text{rad}$$

7) Sidewash Angle for given Yawing Moment Coefficient

$$\text{fx } \sigma = \left(\frac{C_n}{V_v \cdot \eta_v \cdot C_v} \right) - \beta$$

[Open Calculator !\[\]\(b64b40baaee5acddc1eab8538ba84754_img.jpg\)](#)

$$\text{ex } 0.067694\text{rad} = \left(\frac{1.4}{1.02 \cdot 16.66 \cdot 0.7\text{rad}^{-1}} \right) - 0.05\text{rad}$$



8) Sidewash Angle given Yawing Moment Coefficient using Wingspan

$$\text{fx } \sigma = \left(C_n \cdot S \cdot b \cdot \frac{Q_w}{l_v \cdot S_v \cdot Q_v \cdot C_v} \right) - \beta$$

Open Calculator 

ex

$$0.06684\text{rad} = \left(1.4 \cdot 5.08\text{m}^2 \cdot 1.15\text{m} \cdot \frac{0.66\text{Pa}}{1.2\text{m} \cdot 5\text{m}^2 \cdot 11\text{Pa} \cdot 0.7\text{rad}^{-1}} \right) - 0.05\text{rad}$$

9) Yawing Moment Coefficient for given Vertical Tail Lift Curve Slope

$$\text{fx } C_n = l_v \cdot S_v \cdot Q_v \cdot C_v \cdot \frac{\beta + \sigma}{S \cdot b \cdot Q_w}$$

Open Calculator 

ex

$$1.401917 = 1.2\text{m} \cdot 5\text{m}^2 \cdot 11\text{Pa} \cdot 0.7\text{rad}^{-1} \cdot \frac{0.05\text{rad} + 0.067\text{rad}}{5.08\text{m}^2 \cdot 1.15\text{m} \cdot 0.66\text{Pa}}$$

10) Yawing Moment Coefficient for given Vertical Tail Volume Ratio

$$\text{fx } C_n = V_v \cdot \eta_v \cdot C_v \cdot (\beta + \sigma)$$

Open Calculator 

ex

$$1.391743 = 1.02 \cdot 16.66 \cdot 0.7\text{rad}^{-1} \cdot (0.05\text{rad} + 0.067\text{rad})$$

11) Yawing Moment Coefficient using Wingspan

$$\text{fx } C_n = \frac{N_v}{Q_w \cdot S \cdot b}$$

Open Calculator 

ex

$$1.400517 = \frac{5.4\text{N}^*\text{m}}{0.66\text{Pa} \cdot 5.08\text{m}^2 \cdot 1.15\text{m}}$$



Variables Used

- **b** Wingspan (Meter)
- **C_n** Yawing Moment Coefficient
- **C_v** Vertical Tail Lift Curve Slope (1 per Radian)
- **N_v** Vertical Tail Moment (Newton Meter)
- **Q_v** Vertical Tail Dynamic Pressure (Pascal)
- **Q_w** Wing Dynamic Pressure (Pascal)
- **S** Reference Area (Square Meter)
- **S_v** Vertical Tail Area (Square Meter)
- **V_v** Vertical Tail Volume Ratio
- **α_v** Vertical Tail Angle of Attack (Radian)
- **β** Sideslip Angle (Radian)
- **η_v** Vertical Tail Efficiency
- **σ** Sidewash Angle (Radian)
- **l_v** Vertical Tail Moment Arm (Meter)



Constants, Functions, Measurements used

- **Measurement: Length** in Meter (m)
Length Unit Conversion 
- **Measurement: Area** in Square Meter (m^2)
Area Unit Conversion 
- **Measurement: Pressure** in Pascal (Pa)
Pressure Unit Conversion 
- **Measurement: Angle** in Radian (rad)
Angle Unit Conversion 
- **Measurement: Moment of Force** in Newton Meter ($\text{N}\cdot\text{m}$)
Moment of Force Unit Conversion 
- **Measurement: Reciprocal Angle** in 1 per Radian (rad^{-1})
Reciprocal Angle Unit Conversion 



Check other formula lists

- [Aerodynamic Parameters Formulas](#) 
- [Vertical Tail Contribution Formulas](#) 
- [Wing-Tail Interaction Formulas](#) 

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