## Eccentric Loading Formulas

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## List of 18 Eccentric Loading Formulas

## Eccentric Loading

1) Critical Buckling Load given Deflection in Eccentric Loading $\boxed{\Omega}$
$f \times P_{c}=\frac{P \cdot\left(4 \cdot \mathrm{e}_{\text {load }}+\pi \cdot \delta\right)}{\delta \cdot \pi}$
ex $55.41737 \mathrm{kN}=\frac{9.99 \mathrm{kN} \cdot(4 \cdot 2.5 \mathrm{~mm}+\pi \cdot 0.7 \mathrm{~mm})}{0.7 \mathrm{~mm} \cdot \pi}$
2) Cross-Sectional Area given Radius of Gyration in Eccentric Loading
$f \mathrm{x} \mathrm{A}_{\mathrm{cs}}=\frac{\mathrm{I}}{\mathrm{k}_{\mathrm{G}}^{2}}$
ex $13.37693 \mathrm{~m}^{2}=\frac{1.125 \mathrm{~kg} \cdot \mathrm{~m}^{2}}{(0.29 \mathrm{~mm})^{2}}$
3) Cross-Sectional Area given Total Stress is where Load doesn't lie on Plane
$f \mathrm{fx} \mathrm{A}_{\mathrm{cs}}=\frac{\mathrm{P}}{\sigma_{\text {total }}-\left(\left(\frac{\mathrm{e}_{\mathrm{x}} \cdot P \cdot \mathrm{c}_{\mathrm{x}}}{\mathrm{I}_{\mathrm{y}}}\right)+\left(\frac{\mathrm{e}_{\mathrm{y}} \cdot \mathrm{P} \cdot \mathrm{c}_{\mathrm{y}}}{\mathrm{I}_{\mathrm{x}}}\right)\right)}$
ex $13.22767 \mathrm{~m}^{2}=\frac{9.99 \mathrm{kN}}{14.8 \mathrm{~Pa}-\left(\left(\frac{4 \cdot 9.99 \mathrm{kN} \cdot 15 \mathrm{~mm}}{50 \mathrm{k} \cdot \mathrm{m}^{2}}\right)+\left(\frac{0.75 \cdot 9.99 \mathrm{kN} \cdot 14 \mathrm{~mm}}{51 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)\right)}$
4) Cross-Sectional Area given Total Unit Stress in Eccentric Loading
$f x A_{c s}=\frac{P}{f-\left(\left(P \cdot c \cdot \frac{e}{I_{\text {neutral }}}\right)\right)}$
ex $0.532035 \mathrm{~m}^{2}=\frac{9.99 \mathrm{kN}}{100 \mathrm{~Pa}-\left(\left(9.99 \mathrm{kN} \cdot 17 \mathrm{~mm} \cdot \frac{11 \mathrm{~mm}}{23 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)\right)}$
5) Deflection in Eccentric Loading
$\mathrm{fx} \delta=\frac{4 \cdot \mathrm{e}_{\mathrm{load}} \cdot \frac{\mathrm{P}}{\mathrm{P}_{\mathrm{c}}}}{\pi \cdot\left(1-\frac{\mathrm{P}}{\mathrm{P}_{\mathrm{c}}}\right)}$
ex $0.739343 \mathrm{~mm}=\frac{4 \cdot 2.5 \mathrm{~mm} \cdot \frac{9.99 \mathrm{kN}}{53 \mathrm{kN}}}{\pi \cdot\left(1-\frac{9.99 \mathrm{kN}}{53 \mathrm{kN}}\right)}$
6) Distance from XX to outermost fiber given Total Stress where Load doesn't lie on Plane U
$f \mathrm{fx} \mathrm{c}_{\mathrm{y}}=\frac{\left(\sigma_{\text {total }}-\left(\frac{P}{\mathrm{~A}_{\mathrm{cs}}}\right)-\left(\frac{\mathrm{e}_{\mathrm{x}} \cdot P \cdot \mathrm{c}_{\mathrm{x}}}{\mathrm{I}_{\mathrm{y}}}\right)\right) \cdot \mathrm{I}_{\mathrm{x}}}{\mathrm{P} \cdot \mathrm{e}_{\mathrm{y}}}$
ex $13.90997 \mathrm{~mm}=\frac{\left(14.8 \mathrm{~Pa}-\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)-\left(\frac{4 \cdot 9.99 \mathrm{kN} \cdot 15 \mathrm{~mm}}{50 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)\right) \cdot 51 \mathrm{~kg} \cdot \mathrm{~m}^{2}}{9.99 \mathrm{kN} \cdot 0.75}$
7) Distance from YY to outermost fiber given Total Stress where Load doesn't lie on Plane U
$f \mathrm{fx} \mathrm{c}_{\mathrm{x}}=\left(\sigma_{\text {total }}-\left(\left(\frac{\mathrm{P}}{\mathrm{A}_{\mathrm{cs}}}\right)+\left(\frac{\mathrm{e}_{\mathrm{y}} \cdot \mathrm{P} \cdot \mathrm{c}_{\mathrm{y}}}{\mathrm{I}_{\mathrm{x}}}\right)\right)\right) \cdot \frac{\mathrm{I}_{\mathrm{y}}}{\mathrm{e}_{\mathrm{x}} \cdot \mathrm{P}}$
Open Calculator ©

## ex

$14.98345 \mathrm{~mm}=\left(14.8 \mathrm{~Pa}-\left(\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)+\left(\frac{0.75 \cdot 9.99 \mathrm{kN} \cdot 14 \mathrm{~mm}}{51 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)\right)\right) \cdot \frac{50 \mathrm{~kg} \cdot \mathrm{~m}^{2}}{4 \cdot 9.99 \mathrm{kN}}$
8) Eccentricity given Deflection in Eccentric Loading
$f \mathbf{x} \mathrm{e}_{\text {load }}=\left(\pi \cdot\left(1-\frac{\mathrm{P}}{\mathrm{P}_{\mathrm{c}}}\right)\right) \cdot \frac{\delta}{4 \cdot \frac{\mathrm{P}}{\mathrm{P}_{\mathrm{c}}}}$
Open Calculator ©
ex $2.366965 \mathrm{~mm}=\left(\pi \cdot\left(1-\frac{9.99 \mathrm{kN}}{53 \mathrm{kN}}\right)\right) \cdot \frac{0.7 \mathrm{~mm}}{4 \cdot \frac{9.99 \mathrm{kN}}{53 \mathrm{kN}}}$
9) Eccentricity w.r.t axis XX given Total Stress where Load doesn't lie on Plane
$f x e_{y}=\frac{\left(\sigma_{\text {total }}-\left(\frac{P}{A_{c s}}\right)-\left(\frac{e_{x} \cdot P \cdot c_{x}}{I_{y}}\right)\right) \cdot I_{x}}{P \cdot c_{y}}$
Open Calculator ©
$\operatorname{ex} 0.745177=\frac{\left(14.8 \mathrm{~Pa}-\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)-\left(\frac{4 \cdot 9.99 \mathrm{kN} \cdot 15 \mathrm{~mm}}{50 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)\right) \cdot 51 \mathrm{~kg} \cdot \mathrm{~m}^{2}}{9.99 \mathrm{kN} \cdot 14 \mathrm{~mm}}$
10) Eccentricity wrt axis $Y Y$ given Total Stress where Load doesn't lie on Plane
$\mathrm{fx}_{\mathrm{x}} \mathrm{e}_{\mathrm{x}}=\frac{\left(\sigma_{\text {total }}-\left(\frac{\mathrm{P}}{\mathrm{A}_{\mathrm{cs}}}\right)-\frac{\mathrm{e}_{\mathrm{y}} \cdot P \cdot \mathrm{c}_{\mathrm{y}}}{\mathrm{I}_{\mathrm{x}}}\right) \cdot \mathrm{I}_{\mathrm{y}}}{\mathrm{P} \cdot \mathrm{c}_{\mathrm{x}}}$
ex $3.995587=\frac{\left(14.8 \mathrm{~Pa}-\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)-\frac{0.75 \cdot 9.99 \mathrm{kN} \cdot 14 \mathrm{~mm}}{51 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right) \cdot 50 \mathrm{~kg} \cdot \mathrm{~m}^{2}}{9.99 \mathrm{kN} \cdot 15 \mathrm{~mm}}$
11) Load for Deflection in Eccentric Loading
$\mathrm{fx}_{\mathrm{x}}=\frac{\mathrm{P}_{\mathrm{c}} \cdot \delta \cdot \pi}{4 \cdot \mathrm{e}_{\text {load }}+\pi \cdot \delta}$
ex $9.554225 \mathrm{kN}=\frac{53 \mathrm{kN} \cdot 0.7 \mathrm{~mm} \cdot \pi}{4 \cdot 2.5 \mathrm{~mm}+\pi \cdot 0.7 \mathrm{~mm}}$
12) Moment of Inertia about $X X$ given Total Stress where Load doesn't lie on Plane
$f \mathbf{f x} \mathrm{I}_{\mathrm{x}}=\frac{\mathrm{e}_{\mathrm{y}} \cdot \mathrm{P} \cdot \mathrm{c}_{\mathrm{y}}}{\sigma_{\text {total }}-\left(\left(\frac{\mathrm{P}}{\mathrm{A}_{\mathrm{cs}}}\right)+\left(\frac{\mathrm{e}_{\mathrm{x}} \cdot \mathrm{P} \cdot \mathrm{c}_{\mathrm{x}}}{\mathrm{I}_{\mathrm{y}}}\right)\right)}$

$$
\text { ex } 51.33008 \mathrm{~kg} \cdot \mathrm{~m}^{2}=\frac{0.75 \cdot 9.99 \mathrm{kN} \cdot 14 \mathrm{~mm}}{14.8 \mathrm{~Pa}-\left(\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)+\left(\frac{4 \cdot 9.99 \mathrm{kN} \cdot 15 \mathrm{~mm}}{50 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)\right)}
$$

13) Moment of Inertia about $Y Y$ given Total Stress where Load doesn't lie on Plane
$f \mathrm{fx} \mathrm{I}_{\mathrm{y}}=\frac{\mathrm{e}_{\mathrm{x}} \cdot \mathrm{P} \cdot \mathrm{c}_{\mathrm{x}}}{\sigma_{\text {total }}-\left(\left(\frac{\mathrm{P}}{\mathrm{A}_{\mathrm{cs}}}\right)+\left(\frac{\mathrm{e}_{\mathrm{y}} \cdot \mathrm{P} \cdot \mathrm{c}_{\mathrm{y}}}{\mathrm{I}_{\mathrm{x}}}\right)\right)}$
ex $50.05523 \mathrm{~kg} \cdot \mathrm{~m}^{2}=$

$$
\overline{14.8 \mathrm{~Pa}-\left(\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)+\left(\frac{0.75 \cdot 9.99 \mathrm{kN} \cdot 14 \mathrm{~mm}}{51 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)\right)}
$$

14) Moment of Inertia given Radius of Gyration in Eccentric Loading
$f \mathrm{fx}=\left(\mathrm{k}_{\mathrm{G}}^{2}\right) \cdot \mathrm{A}_{\mathrm{cs}}$
ex $1.0933 \mathrm{~kg} \cdot \mathrm{~m}^{2}=\left((0.29 \mathrm{~mm})^{2}\right) \cdot 13 \mathrm{~m}^{2}$
15) Moment of Inertia of Cross-Section given Total Unit Stress in Eccentric Loading
$f x I_{\text {neutral }}=\frac{P \cdot c \cdot e}{f-\left(\frac{P}{A_{c s}}\right)}$
ex $18.82597 \mathrm{~kg} \cdot \mathrm{~m}^{2}=\frac{9.99 \mathrm{kN} \cdot 17 \mathrm{~mm} \cdot 11 \mathrm{~mm}}{100 \mathrm{~Pa}-\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)}$
16) Radius of Gyration in Eccentric Loading

Open Calculator
$f \mathbf{f x} \mathrm{k}_{\mathrm{G}}=\sqrt{\frac{\mathrm{I}}{\mathrm{A}_{\mathrm{cs}}}}$
ex $0.294174 \mathrm{~mm}=\sqrt{\frac{1.125 \mathrm{~kg} \cdot \mathrm{~m}^{2}}{13 \mathrm{~m}^{2}}}$
17) Total Stress in Eccentric Loading when Load doesn't lie on Plane
$f \mathrm{fx} \sigma_{\text {total }}=\left(\frac{\mathrm{P}}{\mathrm{A}_{\mathrm{cs}}}\right)+\left(\frac{\mathrm{e}_{\mathrm{x}} \cdot \mathrm{P} \cdot \mathrm{c}_{\mathrm{x}}}{\mathrm{I}_{\mathrm{y}}}\right)+\left(\frac{\mathrm{e}_{\mathrm{y}} \cdot \mathrm{P} \cdot \mathrm{c}_{\mathrm{y}}}{\mathrm{I}_{\mathrm{x}}}\right)$
ex $14.81323 \mathrm{~Pa}=\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)+\left(\frac{4 \cdot 9.99 \mathrm{kN} \cdot 15 \mathrm{~mm}}{50 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)+\left(\frac{0.75 \cdot 9.99 \mathrm{kN} \cdot 14 \mathrm{~mm}}{51 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)$
18) Total Unit Stress in Eccentric Loading
$\mathrm{fx} \mathrm{f}=\left(\frac{\mathrm{P}}{\mathrm{A}_{\mathrm{cs}}}\right)+\left(\mathrm{P} \cdot \mathrm{c} \cdot \frac{\mathrm{e}}{\mathrm{I}_{\text {neutral }}}\right)$
ex $81.99151 \mathrm{~Pa}=\left(\frac{9.99 \mathrm{kN}}{13 \mathrm{~m}^{2}}\right)+\left(9.99 \mathrm{kN} \cdot 17 \mathrm{~mm} \cdot \frac{11 \mathrm{~mm}}{23 \mathrm{~kg} \cdot \mathrm{~m}^{2}}\right)$

## Variables Used

- $\mathbf{A}_{\mathbf{c s}}$ Cross-Sectional Area (Square Meter)
- c Outermost Fiber Distance (Millimeter)
- $\mathbf{c}_{\mathbf{x}}$ Distance from YY to Outermost Fiber (Millimeter)
- Cy Distance from XX to Outermost Fiber (Millimeter)
- e Distance from Load applied (Millimeter)
- $\mathbf{e}_{\text {Ioad }}$ Eccentricity of Load (Millimeter)
- $\mathbf{e}_{\mathbf{x}}$ Eccentricity with respect to Principal Axis YY
- $e_{y}$ Eccentricity with respect to Principal Axis XX
- f Total Unit Stress (Pascal)
- I Moment of Inertia (Kilogram Square Meter)
- Ineutral Moment of Inertia about Neutral Axis (Kilogram Square Meter)
- $\mathbf{I}_{\mathbf{x}}$ Moment of Inertia about X-Axis (Kilogram Square Meter)
- Iy Moment of Inertia about Y-Axis (Kilogram Square Meter)
- $\mathbf{k}_{\mathbf{G}}$ Radius of Gyration (Millimeter)
- P Axial Load (Kilonewton)
- $\mathbf{P}_{\mathbf{c}}$ Critical Buckling Load (Kilonewton)
- $\boldsymbol{\delta}$ Deflection in Eccentric Loading (Millimeter)
- $\sigma_{\text {total }}$ Total Stress (Pascal)


## Constants, Functions, Measurements used

- Constant: pi, 3.14159265358979323846264338327950288

Archimedes' constant

- Function: sqrt, sqrt(Number)

Square root function

- Measurement: Length in Millimeter (mm)

Length Unit Conversion

- Measurement: Area in Square Meter $\left(\mathrm{m}^{2}\right)$

Area Unit Conversion

- Measurement: Pressure in Pascal (Pa)

Pressure Unit Conversion

- Measurement: Force in Kilonewton (kN)

Force Unit Conversion

- Measurement: Moment of Inertia in Kilogram Square Meter ( $\mathrm{kg} \cdot \mathrm{m}^{2}$ ) Moment of Inertia Unit Conversion


## Check other formula lists

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